1	2	3	★ 4	
				REV
OVERVIEW				

The goal of this document is to present the TCS1 control system in a clear and understandable way and to derive the transfer functions to be used in SIMULINK This document format combines notes, graphs, plots, graphics, etc. all on one page. It is not a traditional report style format.

The method used to present this material was to:

Present a hybrid block diagram of the entire system on one page. The block is not 100% mathematically or functionally correct. It is meant to present the sy
 Explore each block or group of blocks, understand the function, and derive the transfer function for use in SIMULINK.

NOTE: Because of the way the blocks are implemented physically (e.g. inverting summer amplifier) the negative signs on some transfer functions may not show inverting summing amplifier followed by an inverting amplifier with a gain of -1 can be simply shown as a simple summer. SO, keep in mind that if a negative sign inverted into the block or its output will be inverted in the next block.

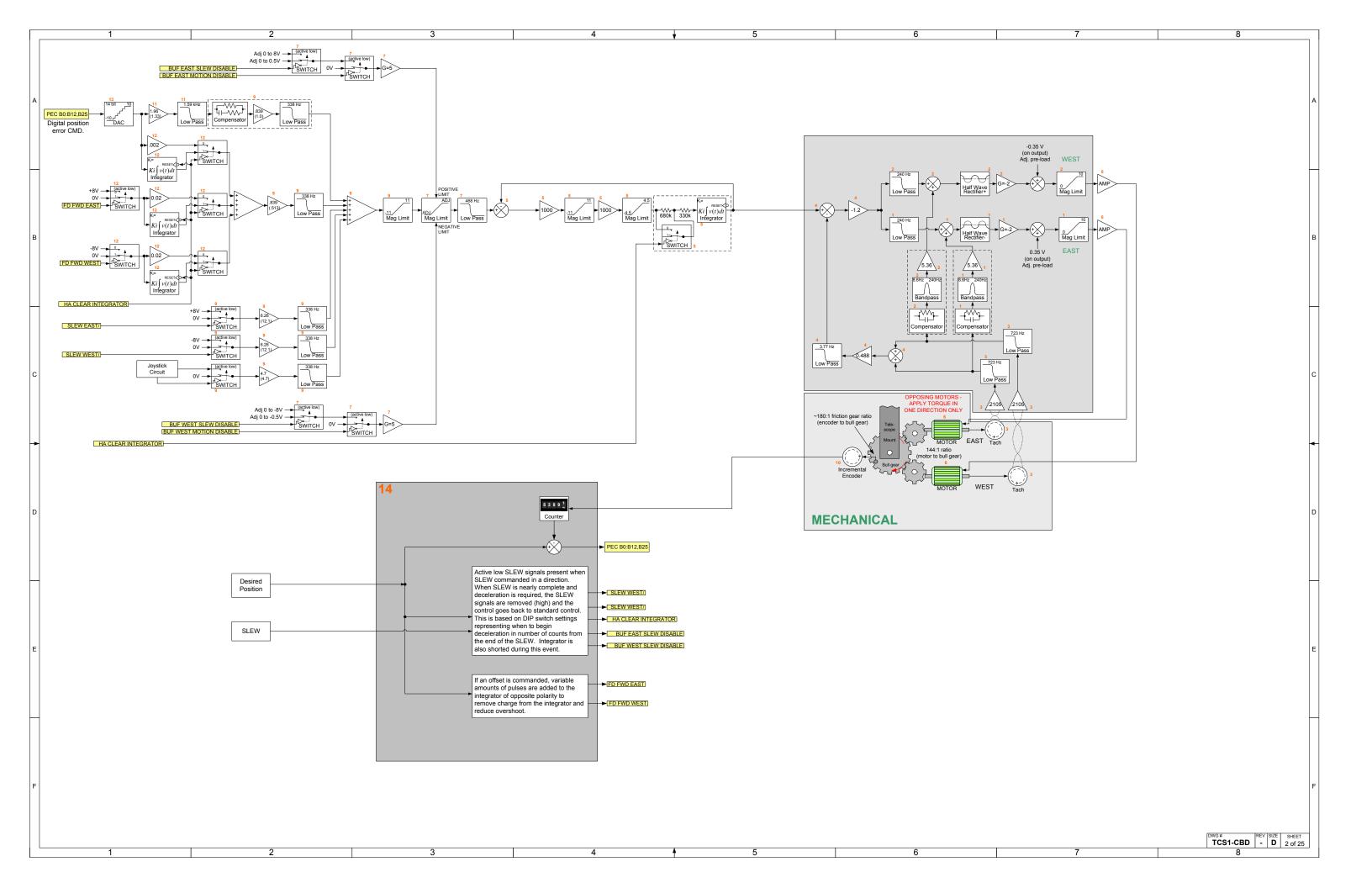
TABLE OF CONTENTS

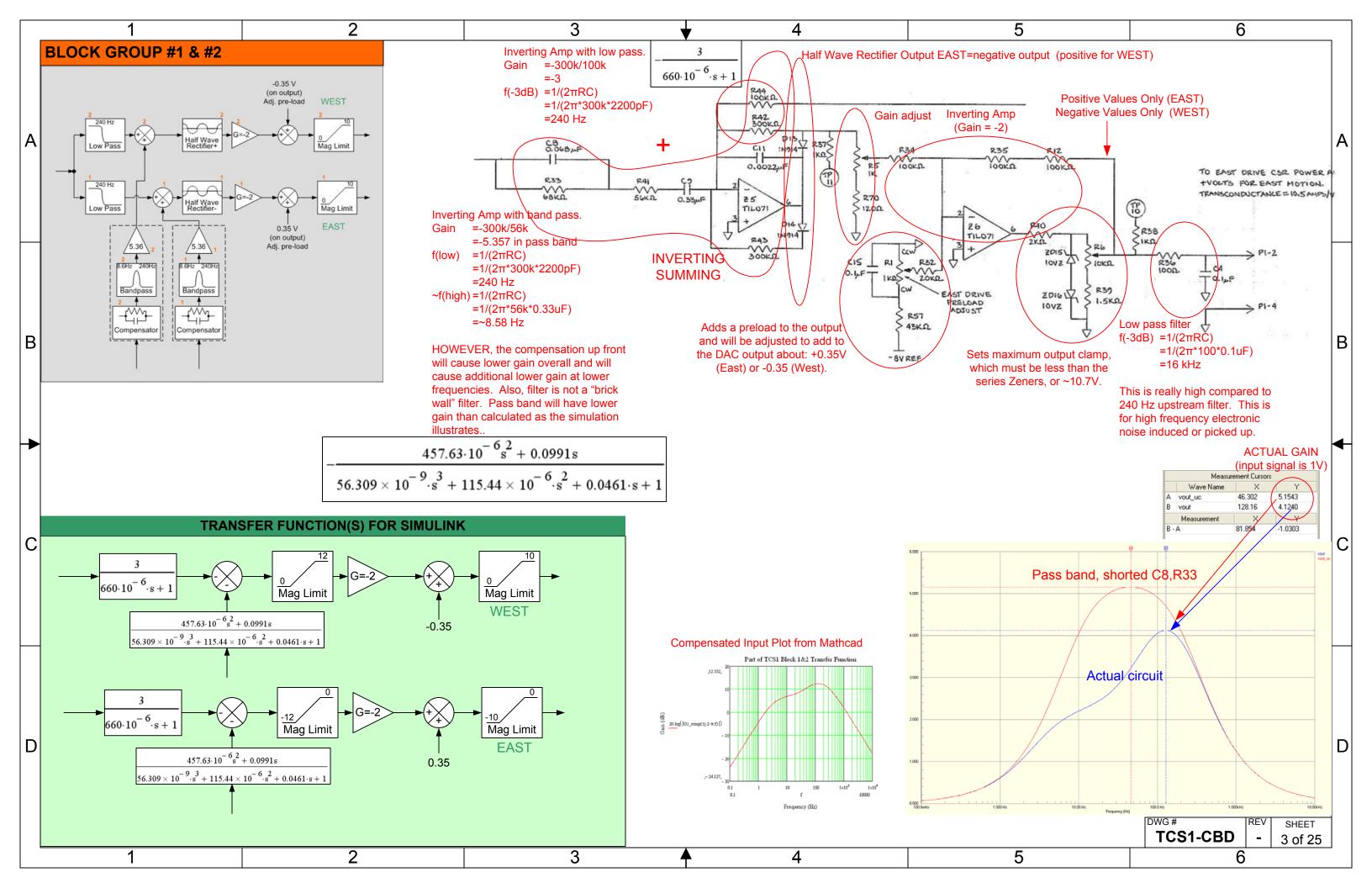
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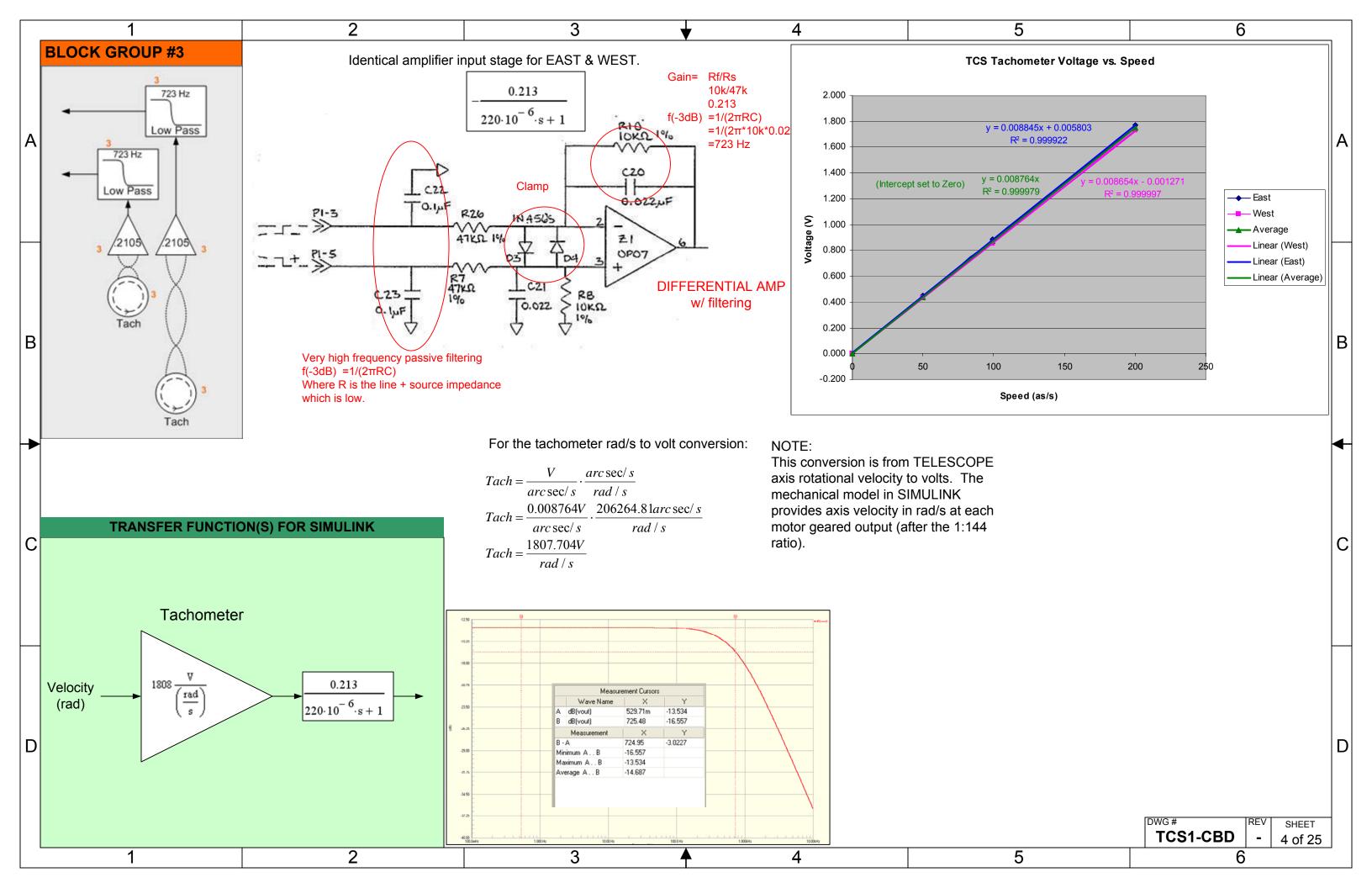
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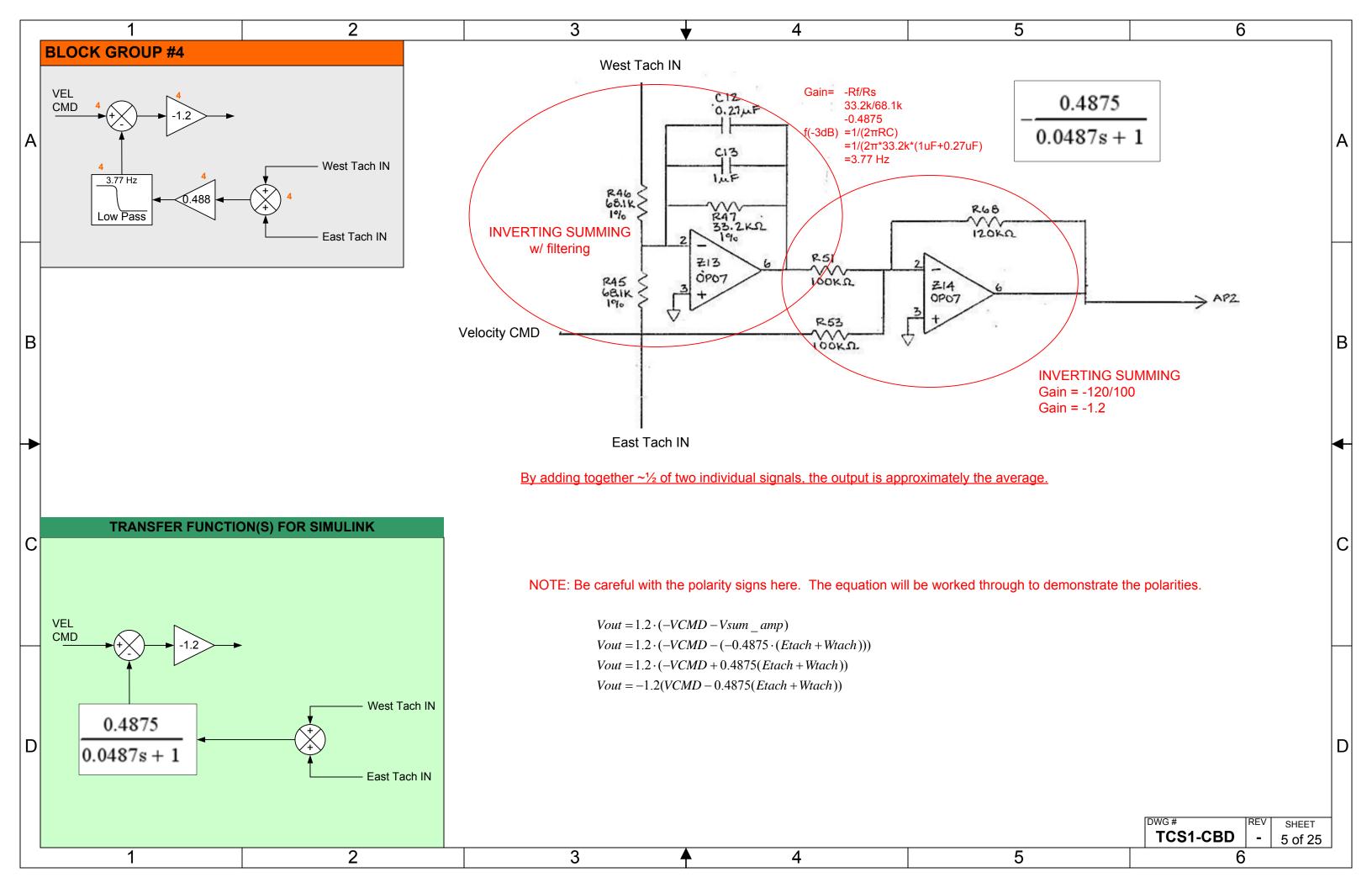
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		<u>PAGE</u>	BLOCK	DESC	CRIPTION			
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		2	-	OVE	RALL BLOCK DIAGRAM			
		3	1 & 2	COM	MAND RECTIFIERS, PRE-LOADS	6, HIGH FREQUENCY TACHOMETE	R COMPENSATION	
		4	3	TACH	IOMETER & TACHOMETER INPL	JT BUFFERS		
ľ		5	4	TACH	HOMETER SUMMER CIRCUIT			
		6	5	ACCI	ELERATION LIMITER			
		7	6	MOT	ORS & BULL GEARS			
		8	6	MOT	OR DATASHEET			
		9	7	COM	MAND MAGNITUDE LIMITER (SL	EW or OFFSET)		
		10	8	POW	ER AMPLIFIER MOTOR DRIVER			
	С	11	9	MAIN	SUMMING BLOCK (PROPORTIC	ONAL GAIN, INTEGRATOR, SLEW FI	EED FORWARD)	
		12	10	INCR	EMENTAL ENCODER & FRICTIC	N GEAR		
		13	11	PRO	PORTIONAL GAIN			
		14	12	INTE	GRATOR WITH SHORTING AND	OFFSET FEED FWD PULSES		
		15	12	OFFS	SET FEED FWD PULSE EXPLANA	ATION		
		16	13	DIGI	TAL TO ANALOG CONVERTER			
		17	13	COM	PUTER/DIGITALLY GENERATED	CONTROL SIGNALS (SLEW, OFFS	ET, POSITION ERROR)	
		18	-	UNIT	CONVERSIONS (SUMMER VOL	TS to RAD/S & RAD/S to ARCSEC/S)		
		19	-	APPE	ENDIX: ABSOLUTE POSITION EN	CODERS		
		20	-	APPE	ENDIX: TCS1 BLOCK DIAGRAM C	CIRCA 1980		
		21	-	APPE	ENDIX: IRTF 123 - TELESCOPE S	ERVO CONTROL HA COMPUTER C	MD & MAGNITUDE DETECT	
		22	-	APPE	ENDIX: IRTF 125 - TELESCOPE S	ERVO CONTROL HA SEQUENCE C	ONTROL BOARD	
	D	23	-	APPE	ENDIX: IRTF 127 & 128 - TELESC	OPE SERVO CONTROL D/A, JOYST	ICK, & INTEGRATOR	
		24	-	APPE	ENDIX: IRTF 130 - TELESCOPE S	ERVO CONTROL TACH SUMMER 8	TORQUE CMD	
		25	-	APPE	ENDIX: IRTF 131 - TELESCOPE S	ERVO CONTROL DEC TACH SUMM	IER & TORQUE CMD	DWG #
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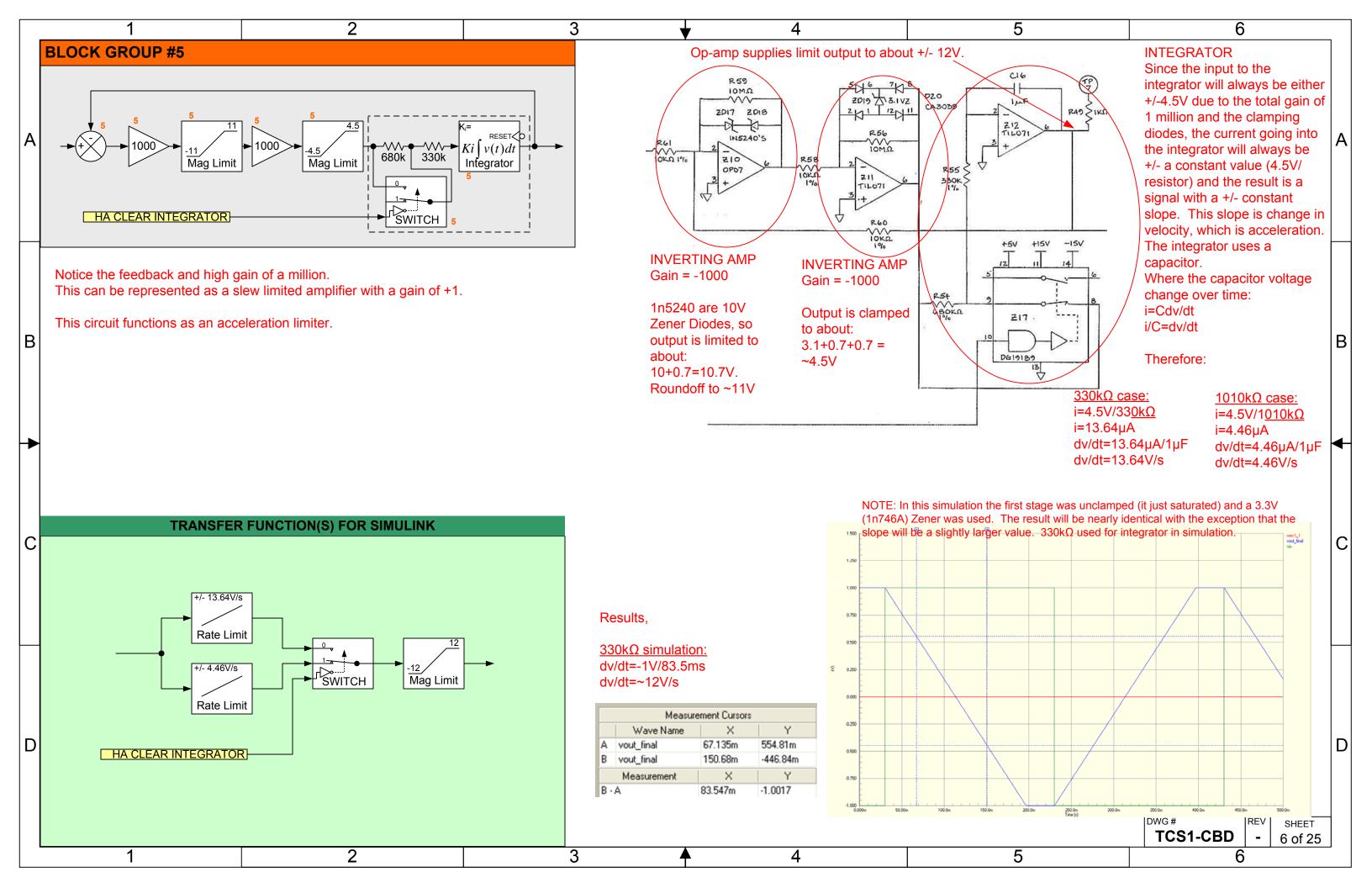
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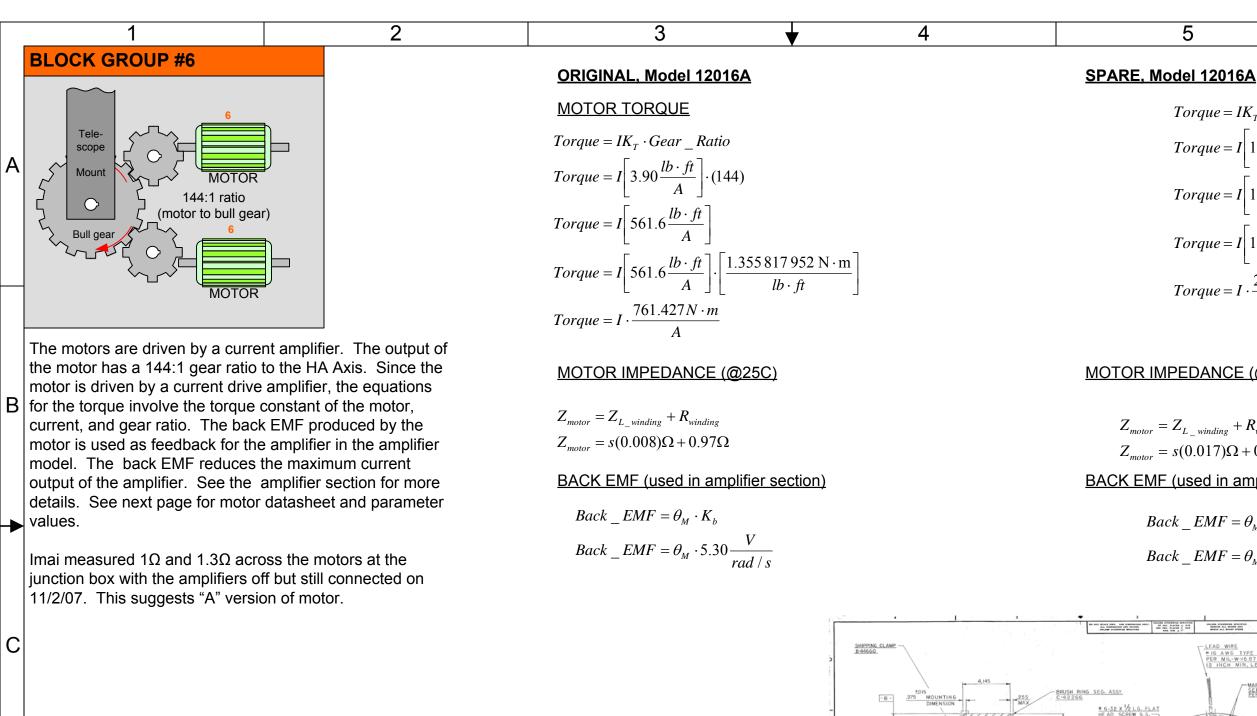


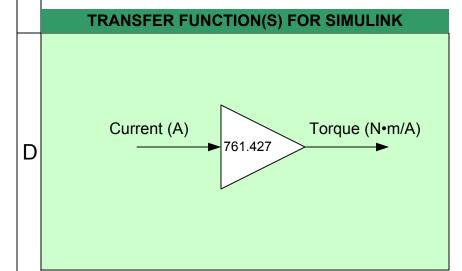




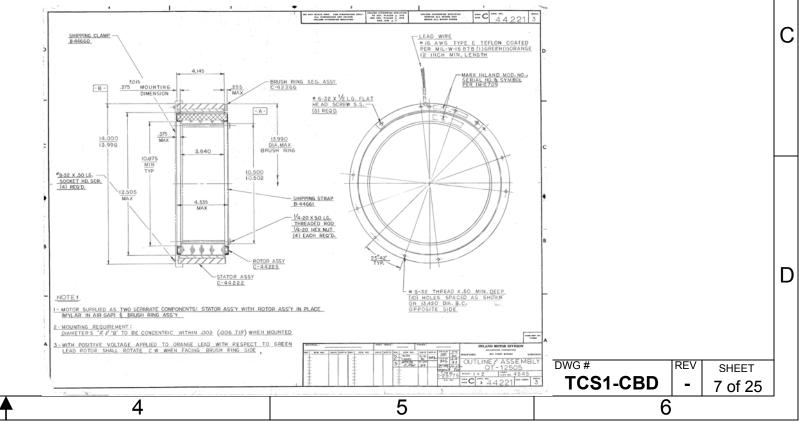








2



5

$$Torque = IK_{T} \cdot Gear _Ratio$$

$$Torque = I\left[12.8 \frac{lb \cdot ft}{A}\right] \cdot (144)$$

$$Torque = I\left[1843.2 \frac{lb \cdot ft}{A}\right]$$

$$Torque = I\left[1843.2 \frac{lb \cdot ft}{A}\right] \cdot \left[\frac{1.355817952 \text{ N} \cdot \text{m}}{lb \cdot ft}\right]$$

$$Torque = I \cdot \frac{2499.04N \cdot m}{A}$$

6

A

В

MOTOR IMPEDANCE (@25C)

$$Z_{L_winding} + R_{winding}$$

$$s(0.017)\Omega + 0.4.50\Omega$$

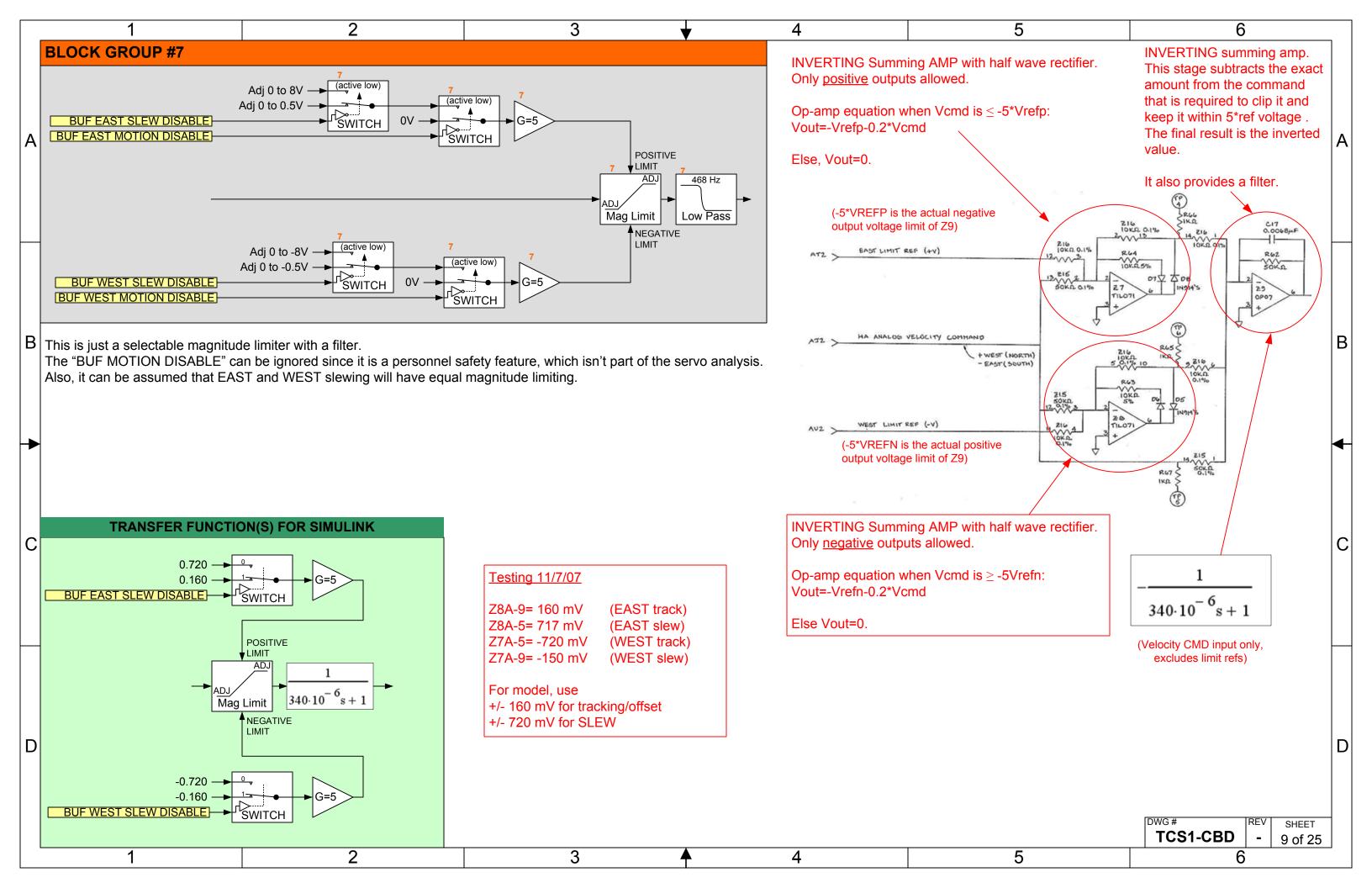
BACK EMF (used in amplifier section)

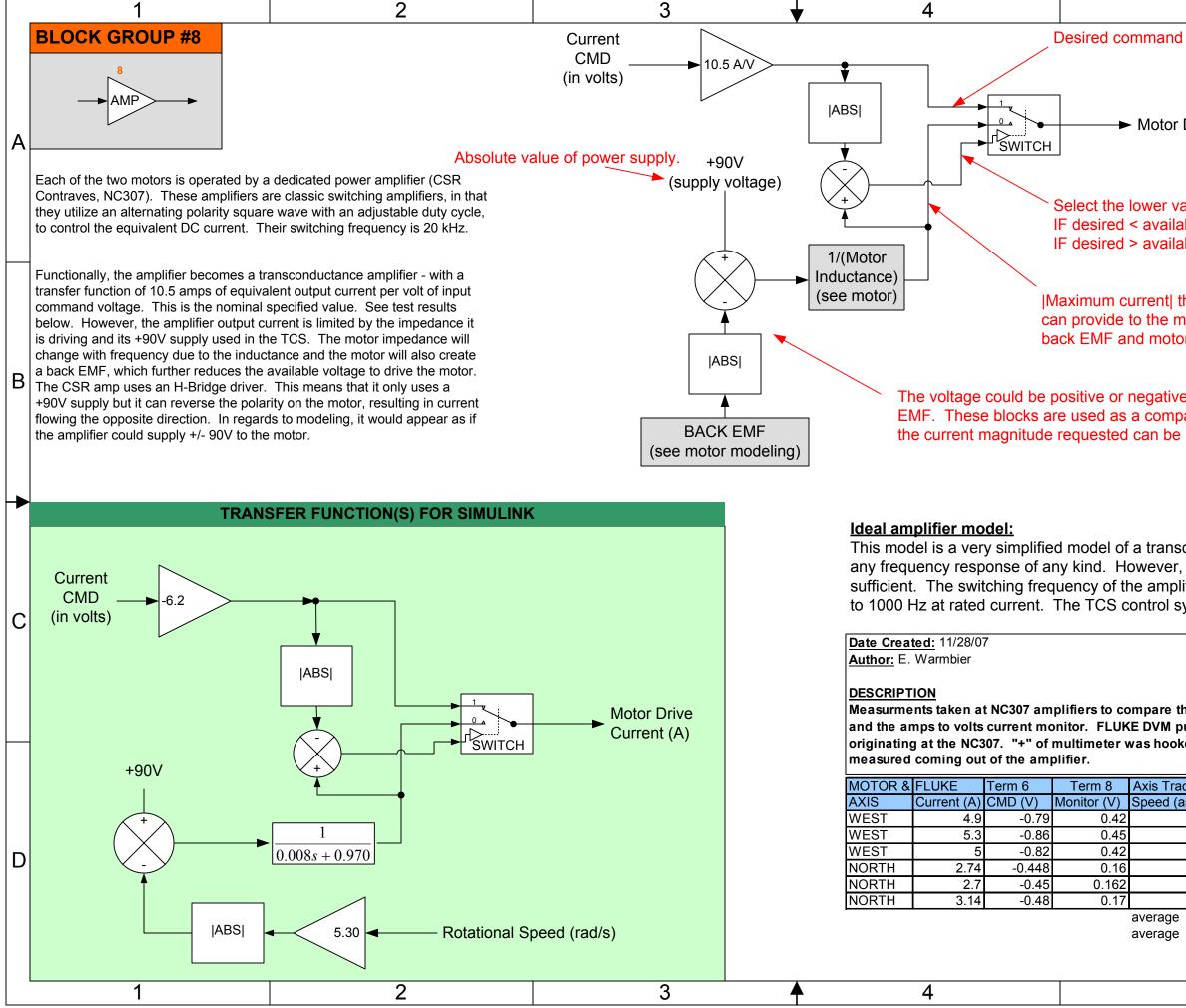
$$k_EMF = \theta_M \cdot K_b$$
$$k_EMF = \theta_M \cdot 17.4 \frac{V}{rad / s}$$

1 BLOCK GROUP #6 CONTINI	2	3	¥	4	5
A Mount Bull gear Bull gear					

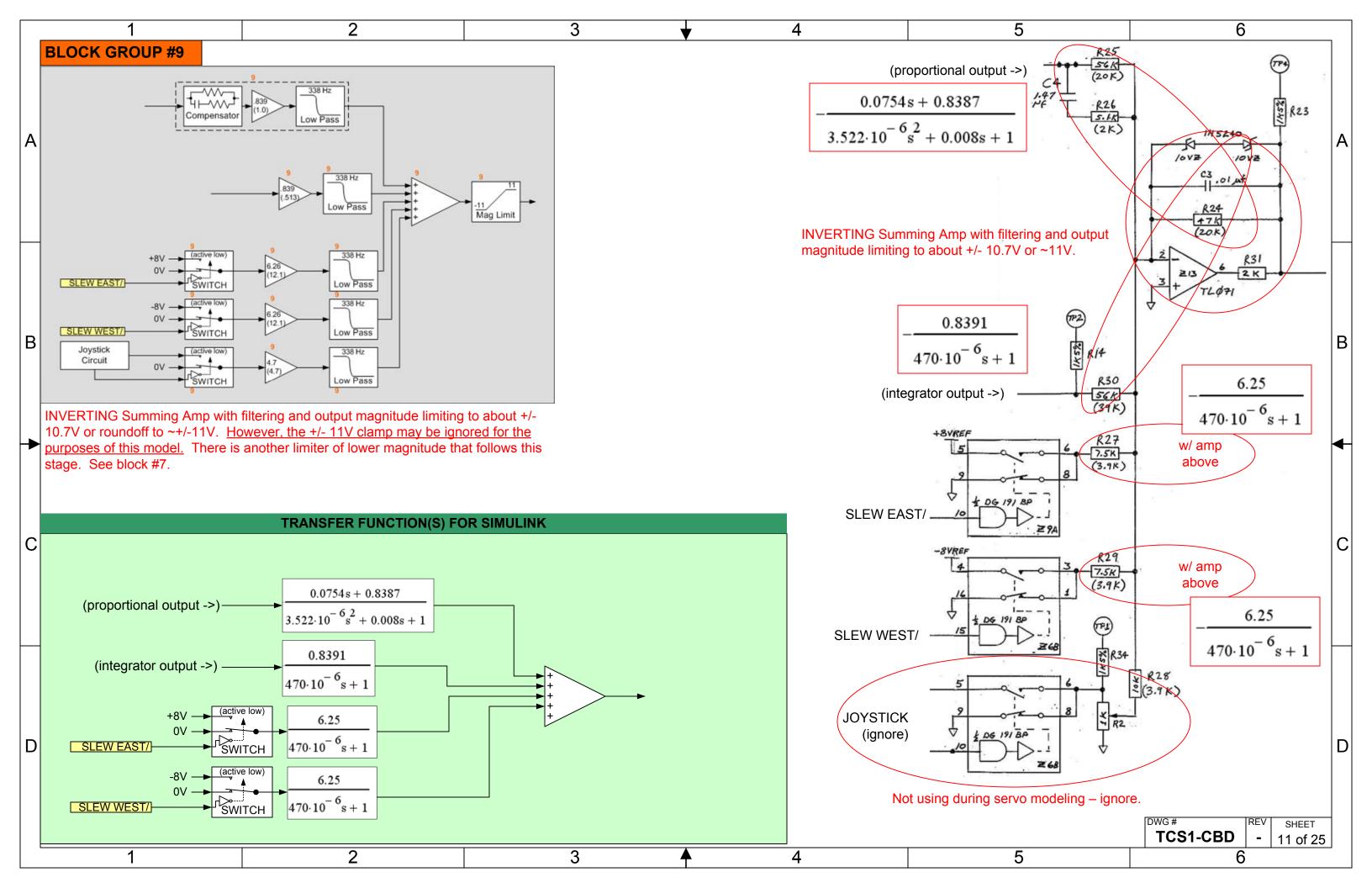
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CKED BY: BPO DATE	E: 03/19/70			MODEL T	12016						ISSUE 2	10		: 10/14/80			MODEL	97-12505						ISSUE B	
DTOR SIZE CONSTANTS	UNITS		SYMBOL	VALUE		Slots	-	99 A	5N(19)	(3)	WD 713	-1 g	MOTOR SIZE CONSTANTS	UNITS		SYNBOL	VALUE		Slots	13	39 /	A 15N20	201	VD 2792	
Peak Torque	lb-ft		Тр	200		Bars		the second s	9817		WD 1118		Peak Torque	lb-ft		Тр	200		Bars		and the second se	B 11N18.		ND 4467	
Notor Constant	Ib-ft/A		Km	3.90		Poles		14 0			MD	5	Hotor Constant	lb-ft/A	1	Km	6.04		Poles	1	28 (c	1.50	MD	
lectical Time Constant	ns		τE	8.50	100 March	M.L. Rot	tor	3.50 D)		WD	TI 5	Electical Time Constant	ns	2.22	TE	3.78		M.L. Roto		3.84 (D .		MD	
lechanical Time Constant	RG		TH	23.0		M.L. Sta	ator	7.31 E		0	WD	- *	Mechanical Time Constant	65		тИ	5.50		M.L. Stat	r	4.15 8	E		MD	
ower In Stalled At Pk Tq 25°C	watts	A	Рр	2600		Ø/Pole	Sec. 15	0 F	F		ND		Power In Stalled At Pk Tq 25°C	watts		Рр	1095		#/Pole	1000	0 1			MD	
liscous Zero Z Source	lb-ft r/s	1000	Fo	21.6		MLT	397.13	1.18 0			WD OL		Viscous Zero 2 Source	lb-ft r	the second s	Fo	49.5		MLT		0.927 (And a state of the		ND	
amping Inf.Z Source	lb-ft r/s		Fi	0.0900)	Rad. Gap	P	0.0500 H	1	2000	WD		Motor Friction Tarque	lb-ft r	/5	Fi	0.30	the second se	Rad. Gap		0.0300 1	н		MD	_
lotor Friction Torque	lb-ft		T#	1.000		No. Brun	sh	8.00		- Spine		-	Ripple Torque Ave. to Peak	percent		TR	2.00	the second se	No. Brush		4.00		13561 544		
tipple Torque Ave. to Peak	percent		TR	2.00		Type Bru	ush	CART S	AME AS T-1	12008 EXCE	EPT HOUNTING	-	Ripple Cycles Per Rev	cycles/			139		Type Brush Brush Are		and the second se	RATING.	-12501 EXC	EPT STACK AN	-
lipple Cycles Per Rev	cycles/rev	1	**	197	100	Brush A	res	0 F	LANGE/ROTO	OR HUB.	1.000	7.000	Ultimate Temp. Rise Per Watt	deg C		TPR	0.10	0	Magnet Ma		18MGO		IP @ 131	CH11 5/A	
Ultimate Temp, Rise Per Watt	deg C		TPR	0.200		Nagnet I	Hat	AL5			ALL ALL ALL ALL		Max Permissible Winding		1.2		0.10		Rotor 0.0	_	12.5		10 131	CHILS/A	
Nax Permissible Winding						Rotor O	.0.	12.0			1	-	Temperature	deg C		1	155		Rotor 1.0		10.5				
Temperature	deg C			105	111	Rotor I.	.0.	99.0		100			Rotor Moment of Inertia	lb-ft-s	1	at	0.27	0	Stator 0.		14.0				-
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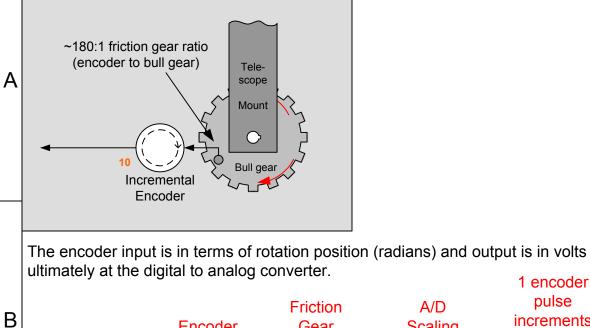


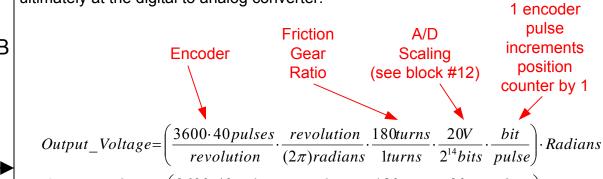


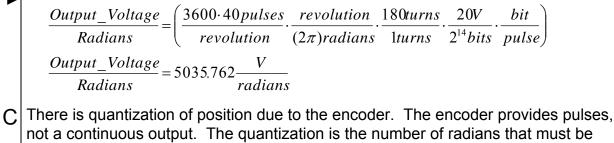
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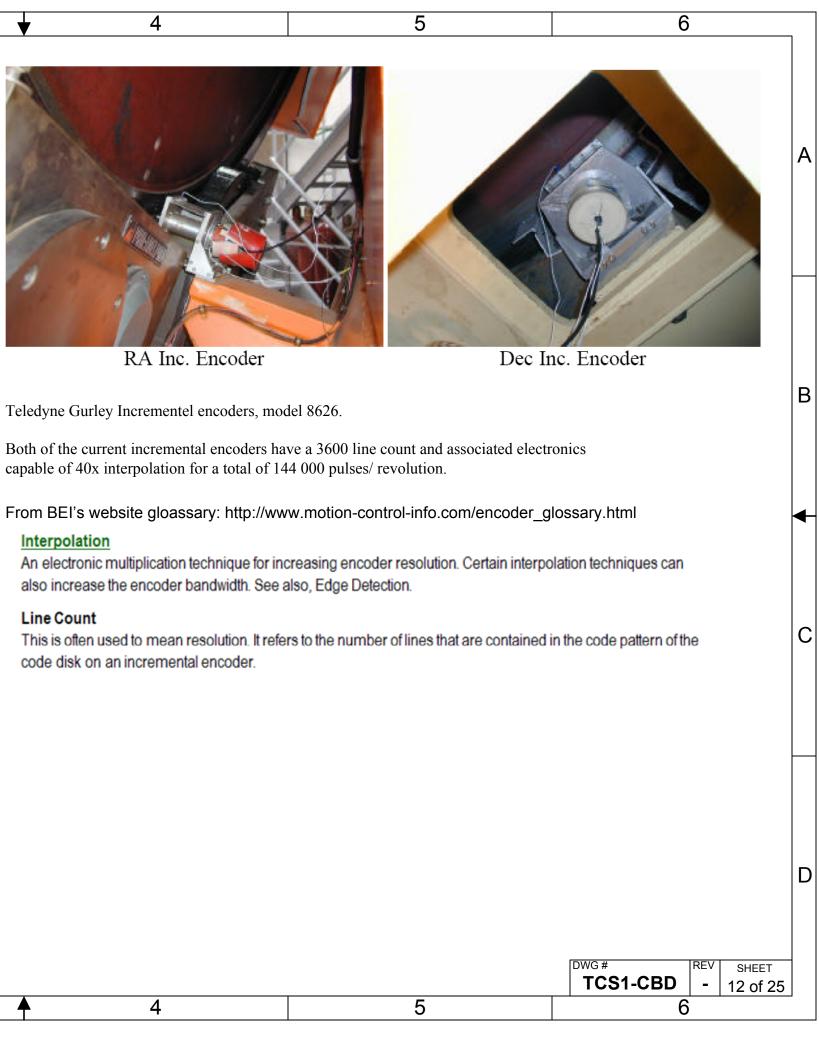




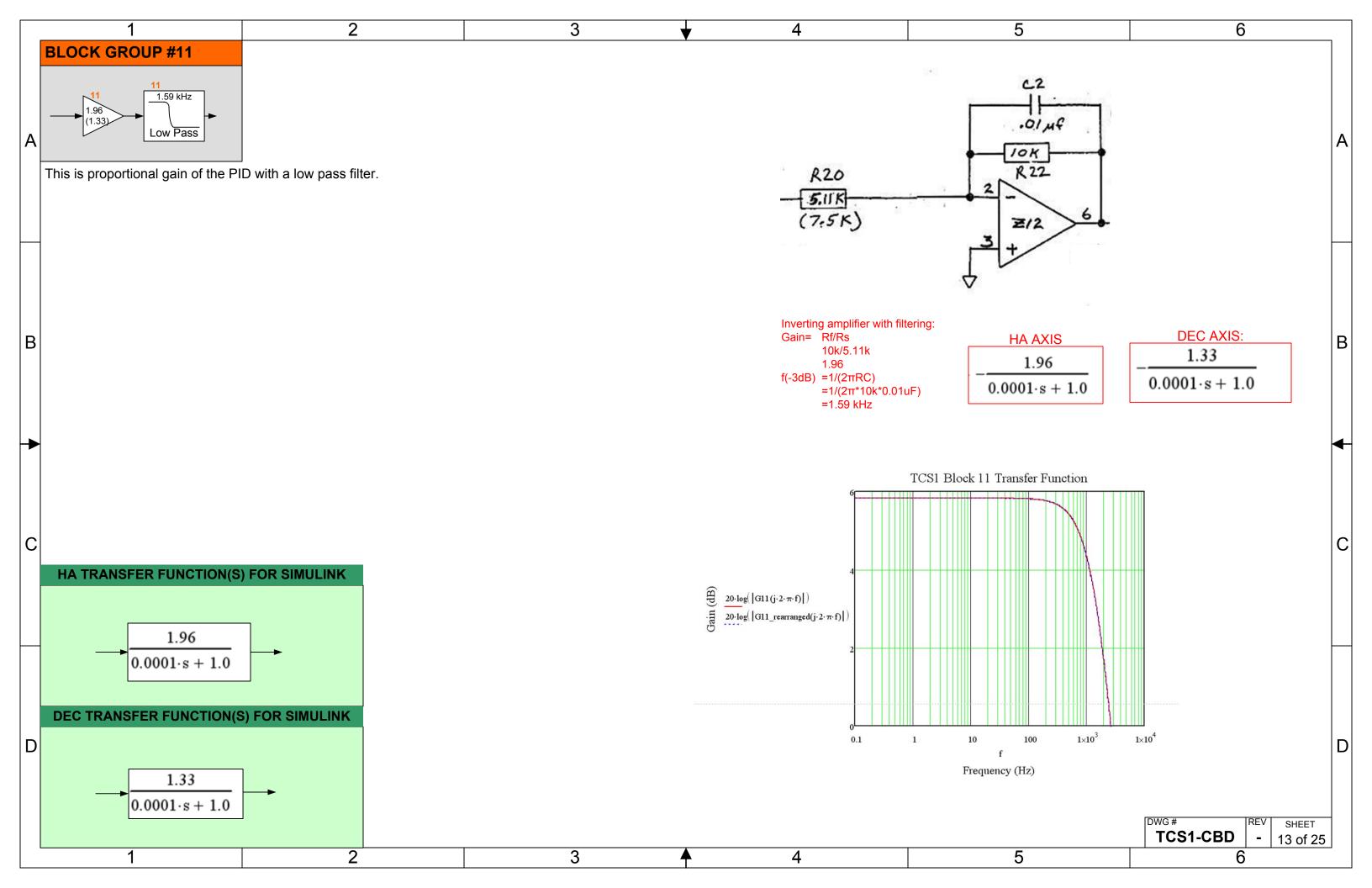


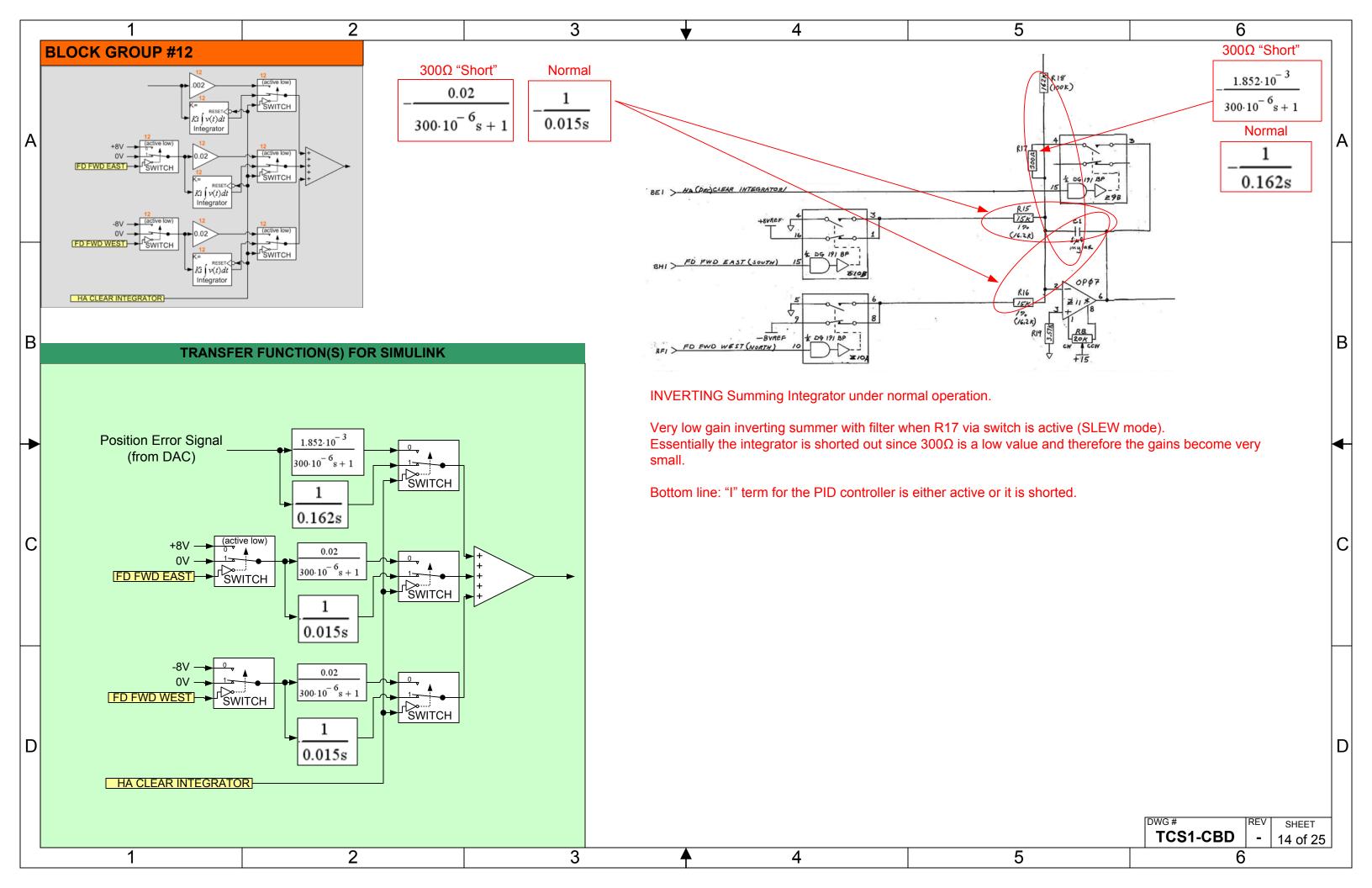


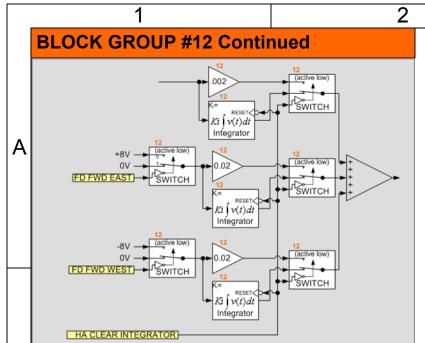
traveled before the next pulse. This is calculated as follows: $\frac{revolution}{3600 \cdot 40 \, pulses} \cdot \frac{(2\pi) radians}{revolution} \cdot \frac{1}{180} = 242.41 \cdot 10^{-9} \frac{radians}{pulse}$ $242.41 \cdot 10^{-9} \frac{radians}{pulse} \cdot 1pulse = 242.41 \cdot 10^{-9} radians$ **TRANSFER FUNCTION(S) FOR SIMULINK** D 5035.76 Encoder 242.41E-9 quantization interval



1	2	3	4	







Below is a portion of the Ev Irwin memo about the offset operation and how the FD FWD is used.

From Ev Irwin Memo:

В

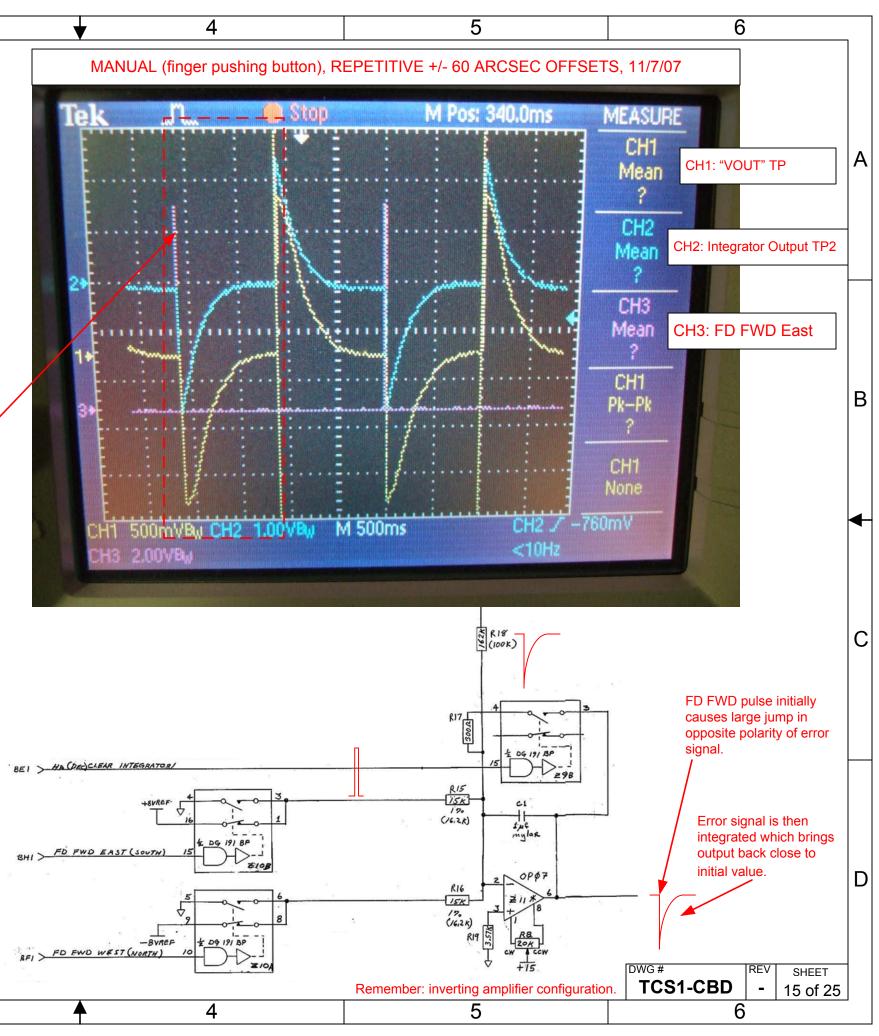
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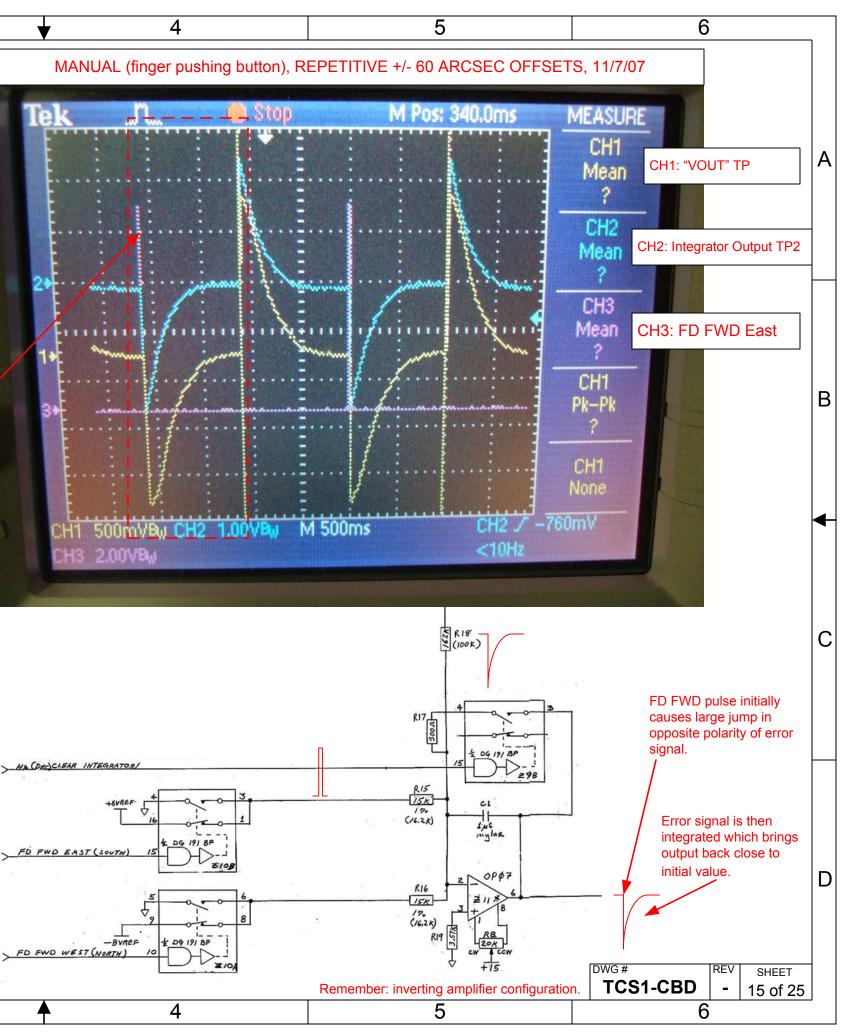
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"In addition to using the PEC D/A signal as an input, during offset or beam switching operations, the Integrator receives pulses of fixed amplitude but varying numbers. These pulses inject charge into the integrator of opposite polarity as is the direction of desired position change. The purpose of these pulses is to reduce overshoot and improve settling time. The number of pulses and therefore total injected charge is one-for-one equal to the commanded number of steps required for relative position change. In other words, when the telescope is commanded to step 400 steps to the east, there is a delay while the telescope moves to the east. During that time, the integrator integrates the error so that when the telescope finally reaches the desired new position, there is excess charge in the integrator. To remove this charge, there would have to be an overshoot with a negative error. The feed-forwarding pulses inject negative charge into the integrator thereby shortening or eliminating the need for overshoot. NOTE - At first, one might wonder why not simply short out the integrator, as is done during slew operations? Unfortunately, during tracking, the integrator contains the amount of charge necessary to produce an output voltage equal to the value required to track the telescope. To short the integrator, would remove that charge, forcing the telescope to briefly stop and thereby worsening the settling time - not improving it."

The scope plot seems to confirm the above statement. When the error reaches zero (CH1) the integrator output (CH2) is approximately the same as the value when the offset started. This is due to the charge injected by the FD FWD pulse (CH3). If this pulse did not exist, the integrator would have a positive value on its output due to integrating the error signal when the error reaches zero.

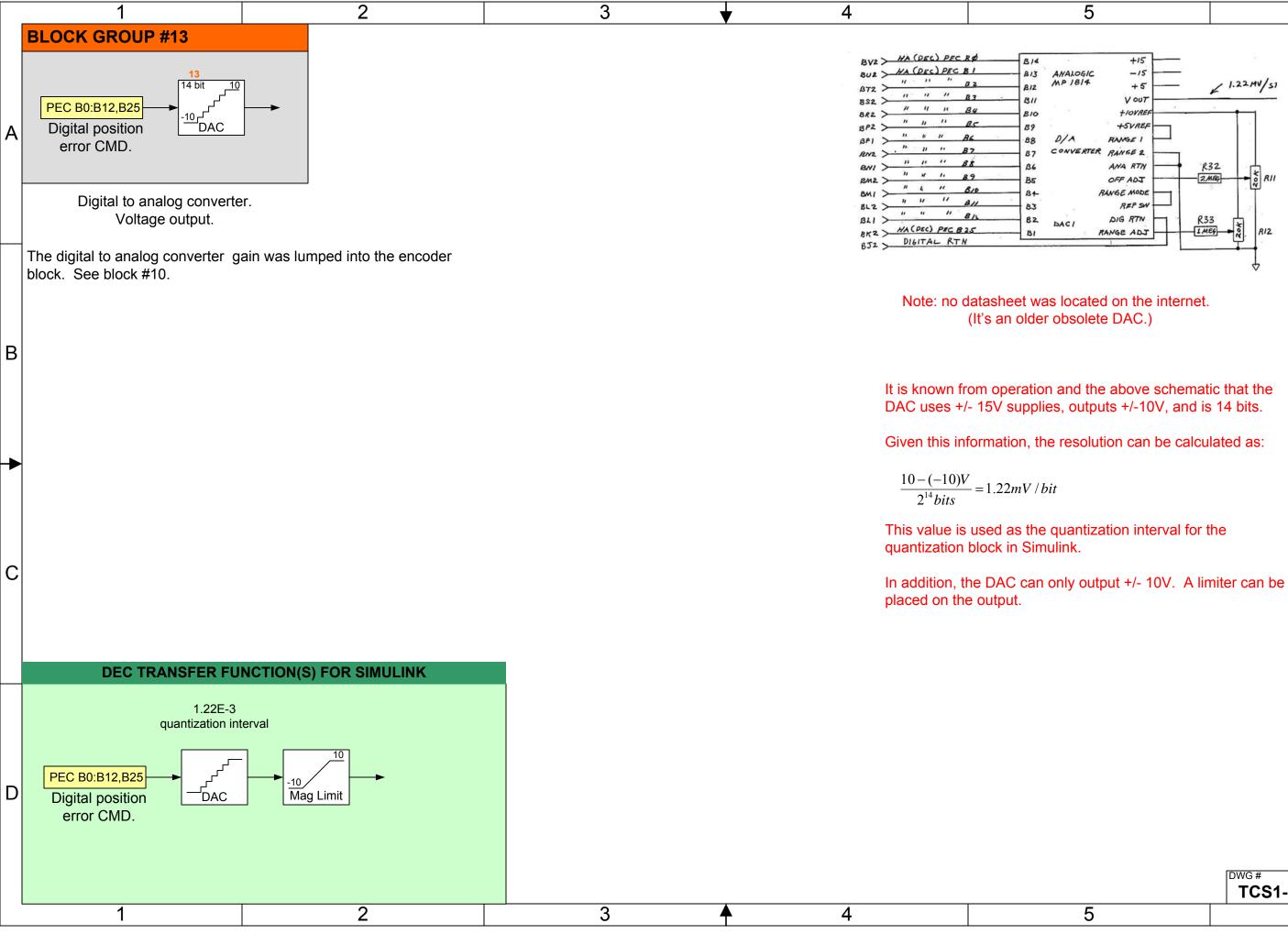
(Keep in mind that the op-amp is inverting.)





2

3



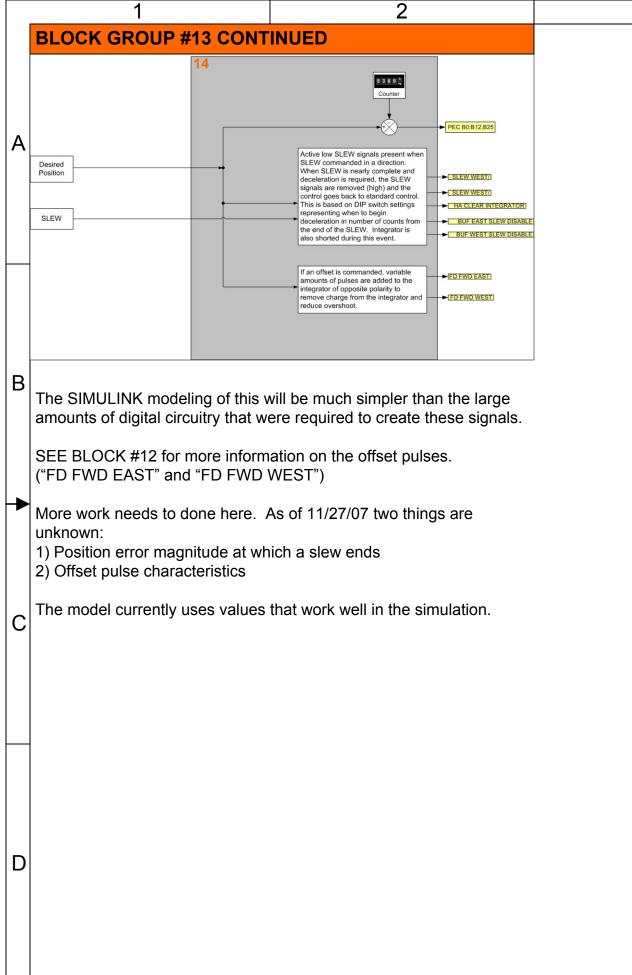
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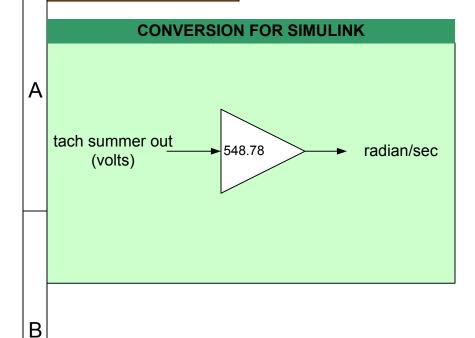
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CONVERSIONS

С

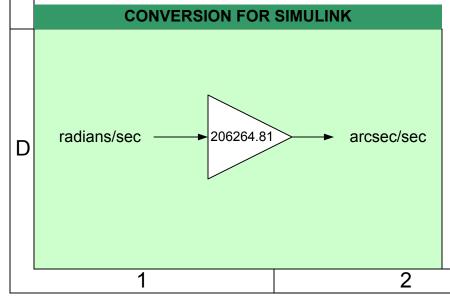


The tachometer summer circuit has a scaled, averaged, tachometer output in volts. In the SIMULINK simulation, it is convenient to display this in arcsec/sec. See conversion derivation for rad/s to arcsec/s used below. This is the feedback that the control system uses.

$$\frac{\operatorname{arc\,sec}}{\operatorname{sec}} = \operatorname{Output}(V) \cdot \frac{1}{\operatorname{Tach_conv}} \cdot \frac{1}{\operatorname{Diff_Amp}} \cdot \frac{1}{\operatorname{summer_gain}} \cdot \frac{1}{\operatorname{Num_tach_input}} \cdot \frac{\operatorname{arc\,sec/s_conversion}}{\operatorname{rad/s}}$$

$$\frac{\operatorname{radians}}{\operatorname{sec}} = \operatorname{Output}(V) \cdot \frac{1}{\frac{1808V}{\operatorname{rad/s}}} \cdot \frac{1}{0.213} \cdot \frac{1}{0.488} \cdot \frac{1}{2} \cdot \frac{206264.81 \operatorname{arc\,sec/s}}{\operatorname{rad/s}}$$

$$\frac{\operatorname{radians}}{\operatorname{sec}} = \operatorname{Output}(V) \cdot \frac{548.78 \operatorname{arc\,sec/s}}{V}$$



In Simulink, some of the outputs will be displayed on a scope. Radians/second are used in the model for radial velocity, but arcsec/sec is what the telescope control displays and uses. Converting it makes it easier to understand and compare to actual telescope data.

radians_	radian	180 deg	60 <i>arc</i> min	60arc sec
sec	sec	$\pi \cdot radian$	deg	arc min
$\frac{radians}{sec} =$	206264	$.81 \frac{arc \sec}{\sec}$		

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APPENDIX: ABSOLUTE POSITION ENCODERS

The current RA absolute position encoder (APE) is located in the North Pier. It is an inductosyn encoder with one half mounted on the yoke, and the other half mounted to the pier.

2

A The DEC APE is located in the west arm of the yoke. It is also an induction type encoder (identical to the RA encoder) with one half mounted to the telescope central section, and the other half mounted on the yoke.

Both of the current APEs have a resolution of 0.1 arcsec.

В

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RA APE

DEC APE

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